

ME 323 Final Examination

Name _____

December 12, 2017

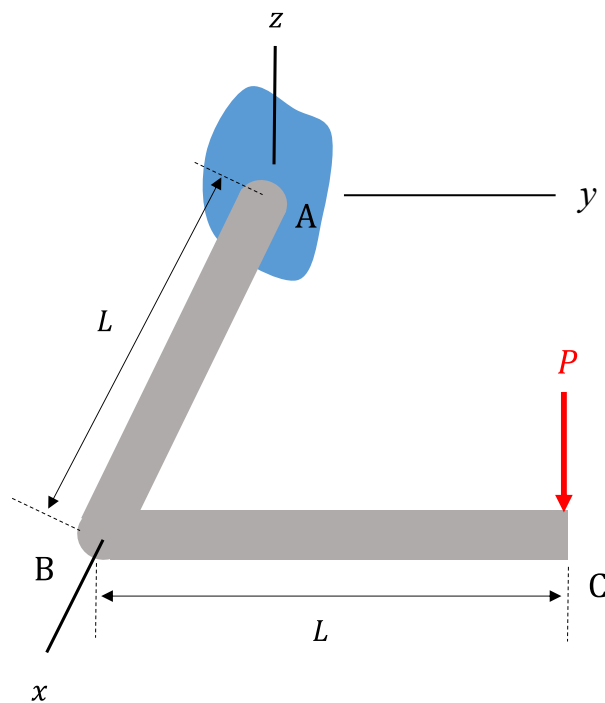
Instructor _____

PROBLEM NO. 1 – 25 points max.

The right-angle bar ABC has a circular cross section of radius r . It is fixed at wall at A, and is subject to a concentrated force P at location C. The elastic modulus of the bar is E and shear modulus is G . Using Castigliano's theorem:

- Determine the deflection of the bar in the z -direction at C.
- Determine the deflection of the bar in the z -direction at B.

Ignore the shear energy due to bending. Express your answers in terms of P , L , E , G , and r .

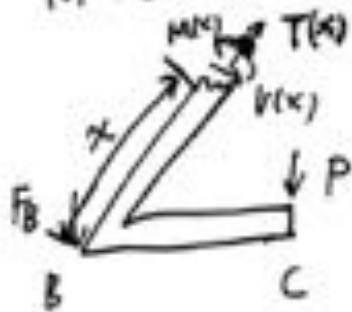


P1: Solution:

Apply a dummy force at B: F_B

$$U_{total} = U_{AB} + U_{BC}$$

For AB section:

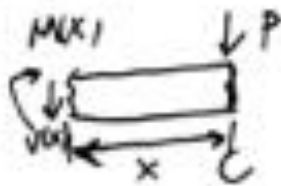


$$T(x) = -PL$$

$$M(x) = (P + F_B)x$$

$$U_{AB} = \frac{1}{2} \int_0^L \frac{T^2(x)}{GIp} dx + \frac{1}{2} \int_0^L \frac{M^2(x)}{EI} dx$$
$$= \frac{1}{2} \int_0^L \frac{(PL)^2}{GIp} dx + \frac{1}{2} \int_0^L \frac{[(P + F_B)x]^2}{EI} dx$$

For BC section:



$$M(x) = -Px$$

$$U_{BC} = \frac{1}{2} \int_0^L \frac{M^2(x)}{EI} dx = \frac{1}{2} \int_0^L \frac{(Px)^2}{EI} dx$$

$$\Delta_C = \frac{\partial U_{total}}{\partial P} \Big|_{F_B=0} = \frac{PL^3}{GIp} + \frac{\frac{1}{3}PL^3}{EI} + \frac{1}{3} \frac{PL^3}{EI} = \frac{PL^3}{GIp} + \frac{2}{3} \frac{PL^3}{EI}$$

$$\Delta_B = \frac{\partial U_{total}}{\partial F_B} \Big|_{F_B=0} = \frac{1}{3} \frac{PL^3}{EI}$$

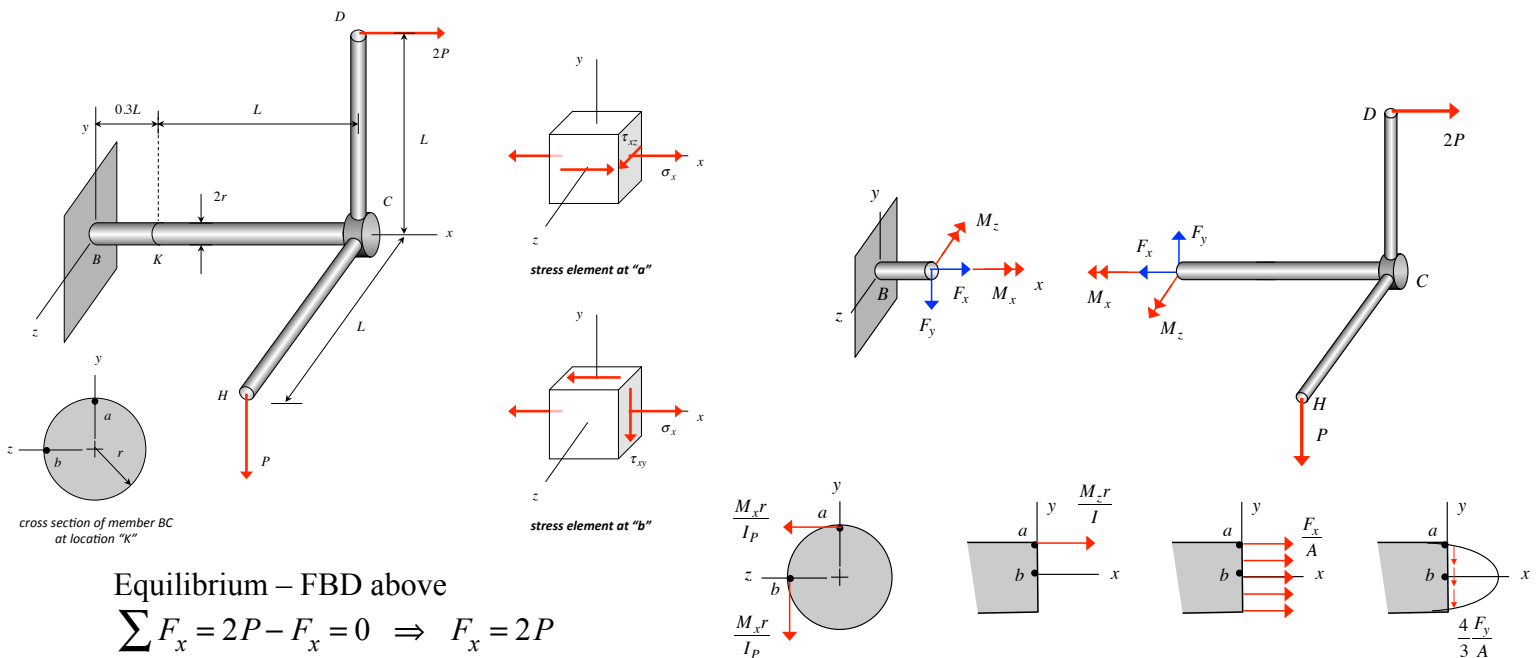
$$I_p = \frac{\pi r^4}{2}, \quad I = \frac{\pi r^4}{4}$$

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PROBLEM NO. 2 – 25 points max.

A bracket is made up of components BC, DC and HC, all joined together at C, with BC, DC and HC aligned with the x-, y- and z-axes, respectively. Loads of P and 2P act at ends H and D in the negative y-direction and positive x-direction, respectively. Component BC has a circular cross section with an outer radius of r.

- Determine the xyz components of stress at points “a” and “b” on the cross section K of component BC. Express your answers in terms of, at most, L, r and P.
- Show these components of stress on the stress elements provided below for points “a” and “b”.



Equilibrium – FBD above

$$\sum F_x = 2P - F_x = 0 \Rightarrow F_x = 2P$$

$$\sum F_y = -P + F_y = 0 \Rightarrow F_y = P$$

$$\sum \vec{M}_K = (\hat{L}_i + \hat{L}_k) \times (-P\hat{j}) + (\hat{L}_i + \hat{L}_j) \times (2P\hat{i}) - M_x\hat{i} - M_z\hat{k} = (PL - M_x)\hat{i} + (-PL - 2PL + M_z)\hat{k} = \vec{0} \Rightarrow$$

$$\hat{i} : M_x = PL$$

$$\hat{k} : M_z = 3PL$$

Using: $I = \frac{\pi}{4}r^4$, $I_p = \frac{\pi}{2}r^4$ and $A = \pi r^2$, and the stress distributions above, we have:

Point “a”

$$\sigma_x = \frac{M_z r}{I} + \frac{F_x}{A} = \frac{(3PL)r}{\frac{\pi r^4}{4}} + \frac{2P}{\pi r^2} = \frac{2P}{\pi r^2} \left[1 + 6 \frac{L}{r} \right]$$

$$\tau_{xz} = \frac{M_x r}{I_p} = \frac{(PL)r}{\frac{\pi r^4}{2}} = \frac{2P}{\pi r^2} \left(\frac{L}{r} \right)$$

Point “b”

$$\sigma_x = \frac{F_x}{A} = \frac{2P}{\pi r^2}$$

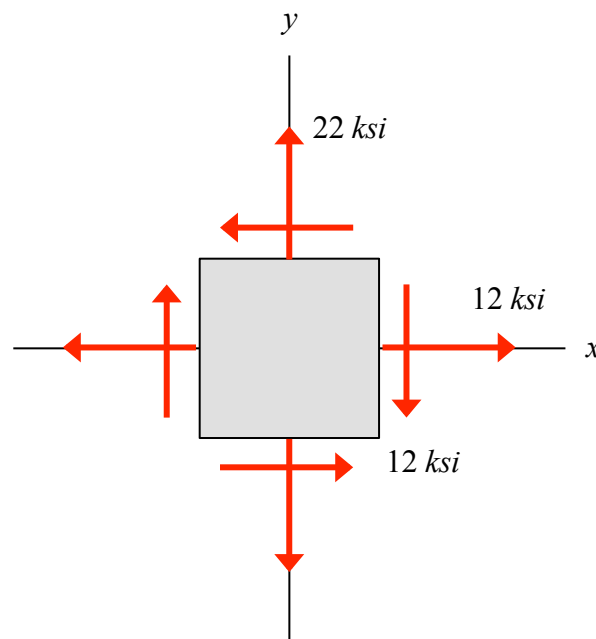
$$\tau_{xy} = -\frac{M_x r}{I_p} - \frac{4 F_y}{3 A} = -\frac{(PL)r}{\frac{\pi r^4}{2}} - \frac{4 P}{3 \pi r^2} = -\frac{P}{\pi r^2} \left(2 \frac{L}{r} + \frac{4}{3} \right)$$

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PROBLEM NO. 3A – 15 points max.

The x-y components for a state of plane stress in a machine component are shown below. The component is made up from a ductile material having a yield strength of $\sigma_y = 36 \text{ ksi}$. For this state of stress:

- Determine if the *maximum-shear stress theory* predicts a failure of the material. If the material has not failed, what is the factor of safety predicted by this theory?
- Determine if the *maximum-distortional-energy theory* predicts a failure of the material. If the material has not failed, what is the factor of safety predicted by this theory?



$$\sigma_{ave} = \frac{12 + 22}{2} = 17 \text{ ksi}$$

$$R = \sqrt{\left(\frac{12 - 22}{2}\right)^2 + (-12)^2} = 13 \text{ ksi}$$

$$\therefore \begin{cases} \sigma_{p1} = \sigma_{ave} + R = 30 \text{ ksi} \\ \sigma_{p2} = \sigma_{ave} - R = 4 \text{ ksi} \end{cases}$$

Since both σ_{p1} & σ_{p2} are positive \Rightarrow

$$|\tau|_{max,abs} = \frac{\sigma_{p1}}{2} = 15 \text{ ksi}$$

$$(a) FS_{MSS} = \frac{\sigma_Y / 2}{|\tau|_{\max, abs}} = \frac{36 / 2}{15} = 1.20$$

$$(b) \tau_m = \sqrt{\sigma_{p1}^2 - \sigma_{p1} \sigma_{p2} + \sigma_{p2}^2}$$
$$= \sqrt{30^2 - (30)(4) + 4^2}$$
$$= 28.2 \text{ ksi}$$

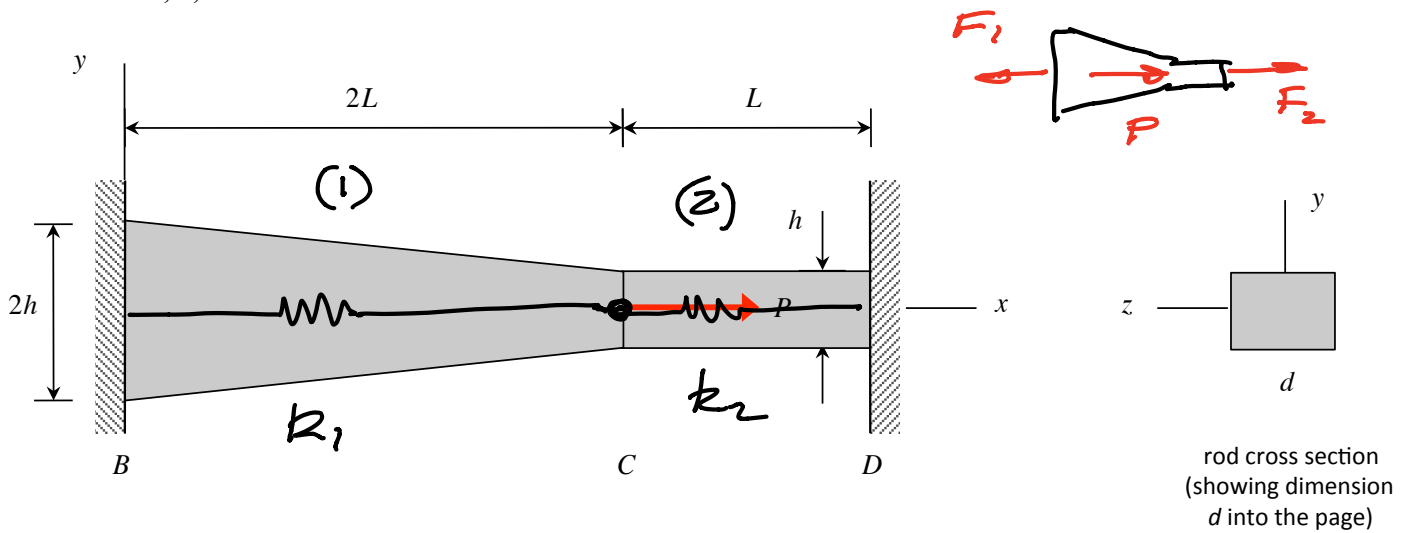
$$\therefore FS_{MDE} = \frac{\sigma_Y}{\tau_m} = \frac{36}{28.2} = 1.28$$

(Note: $FS_{MDE} > FS_{MSS}$, as expected)

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PROBLEM NO. 3B - 10 points max.

A rectangular cross-section rod is made up of sections BC and CD, with each section having a constant depth dimension d as shown in the figures below and being made up of a material having a Young's modulus of E . Section BC has a thickness that varies linearly from $2h$ on its left end to h on its right end at C, whereas section CD has a constant thickness h throughout its length. Ends B and D of the rod are connected to rigid walls. An axial load P acts the point C where the two sections are joined. Use a two-element *finite element model* to determine the *stress experienced in section CD of the rod*. Express your answer in terms of, at most, the variables h, d, L and E .



$$k_1 = \frac{E}{2L} A_{1,ave} = \frac{E}{2L} \frac{1}{2} (2hd + hd) = \frac{3}{4} \frac{Ehd}{L}$$

$$k_2 = \frac{E}{L} A_{2,ave} = \frac{Ehd}{L}$$

$$[K] = \begin{bmatrix} k_1 & -k_1 & 0 \\ -k_1 & k_1 + k_2 & -k_2 \\ 0 & -k_2 & k_2 \end{bmatrix} = \begin{bmatrix} \frac{3}{4} & -\frac{3}{4} & 0 \\ -\frac{3}{4} & \frac{7}{4} & -1 \\ 0 & -1 & 1 \end{bmatrix} \frac{Ehd}{L}$$

$$\{F\} = \begin{Bmatrix} -F_1 \\ P \\ F_2 \end{Bmatrix}$$

$$[K]\{u\} = \{F\}$$

To enforce BC's, remove 1st/3rd columns and rows of $[K]$ and the 1st/3rd rows of $[F]$:

$$\left(\frac{7}{4} \frac{Ehd}{L}\right) u_2 = P \Rightarrow u_2 = \frac{4PL}{7Ehd}$$

From 3rd equation prior to enforcing BC's:

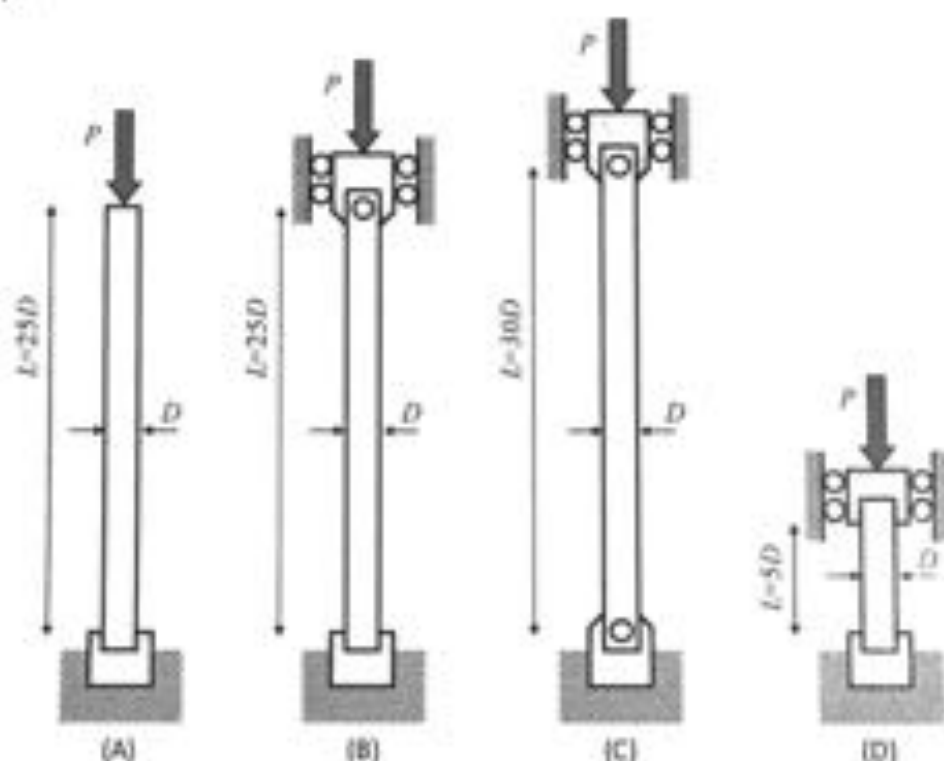
$$-\frac{Ehd}{L} u_2 + \frac{Ehd}{L} u_3 = F_2$$

$$\hookrightarrow F_2 = -\frac{Ehd}{K} \left(\frac{4PK}{7Ehd}\right) = -\frac{4}{7} P$$

$$\therefore \sigma_2 = \frac{F_2}{A_2} = \frac{-\frac{4}{7} P}{hd} = \ominus \frac{4}{7} \frac{P}{hd}$$

↓ compression

PROBLEM NO. 4A - 5 points max. (partial credit, please show your work here and on the next page):



Cylindrical columns A, B, C and D shown above are made of the same material (Young's modulus of E and yield stress of $\sigma_y = E/100$) and have the same circular cross-section of diameter D . A compressive axial load P is applied to each column.

Let P_{cr}^A , P_{cr}^B , P_{cr}^C , and P_{cr}^D represent the critical buckling loads for columns A, B, C, and D, respectively.

Rank order (from largest to smallest, 1 to 5) the following: P_{cr}^A , P_{cr}^B , P_{cr}^C , P_{cr}^D , and $\pi R^2 \sigma_y / 2$ below.

$$P_{cr}^A \quad \underline{5}$$

$$P_{cr}^B \quad \underline{3}$$

$$P_{cr}^C \quad \underline{4}$$

$$P_{cr}^D \quad \underline{1}$$

$$\frac{\pi R^2 \sigma_y}{2} \quad \underline{2}$$

(A) $l_{eff} = 2L$

$S_r = \frac{BL}{D} = 200 > (S_r)_{cr} \Rightarrow$ Euler's column

$P_{cr}^A = \frac{\pi EI}{l_{eff}^2} = \frac{\pi EI}{200^2}$

$S_r = \frac{l_{eff}}{\sqrt{I/A}} = \frac{4l_{eff}}{D}$

$(S_r)_{cr} = \pi \sqrt{\frac{2E}{\sigma_T}} = 44.4$

(B) $l_{eff} = 0.7L$

$\Rightarrow S_r = \frac{2.8L}{D} = 70 > (S_r)_{cr} \Rightarrow$ Euler's column

$P_{cr}^B = \frac{\pi EI}{l_{eff}^2} = \frac{\pi EI}{70^2}$

(C) $l_{eff} = L$

$\Rightarrow S_r = \frac{4L}{D} = 120 > (S_r)_{cr} \Rightarrow$ Euler's column

$P_{cr}^C = \frac{\pi EI}{l_{eff}^2} = \frac{\pi EI}{120^2}$

(D) $l_{eff} = 0.7L$

$\Rightarrow S_r = \frac{2.8L}{D} = 14 < (S_r)_{cr} \Rightarrow$ Johnson's column.

$P_{cr}^D = \sigma_Y A \left[1 - \frac{\sigma_Y S_r^2}{4\pi^2 E} \right] = 0.95 \sigma_Y A > \sigma_Y \frac{\pi R^2}{2}$

Also, for given E and σ_Y , any Euler's column will have a buckling load smaller than the buckling load of a Johnson's column.

$[at (S_r)_{cr} P_{cr} = \sigma_Y A / 2]$

ORDER :

$P_{cr}^D > \frac{\sigma_Y \pi R^2}{2} > P_{cr}^B > P_{cr}^C > P_{cr}^A$

$\Rightarrow P_{cr}^D > \frac{\sigma_Y \pi R^2}{2}$
 $P_{cr}^D > P_{cr}^A$
 $P_{cr}^D > P_{cr}^B$
 $P_{cr}^D > P_{cr}^C$

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The schematics presented below correspond to beams with span L and loaded with either a couple at the left end or with a force per unit of length with different distributions (i.e., linear, constant or quadratic).

Problem 4B - 7 points max. (partial credit, please show your work on next page)

Which schematic corresponds to a beam whose bending moment is equal to

$$M(x) = \frac{x w_0 L}{10} - \frac{w_0 x^2}{6 L}$$

Circle the correct answer (a) (b) (c) (d) (e) (f) (g) (h) (i) (j)



Problem 4C - 7 points max. (partial credit, please show your work on next page)

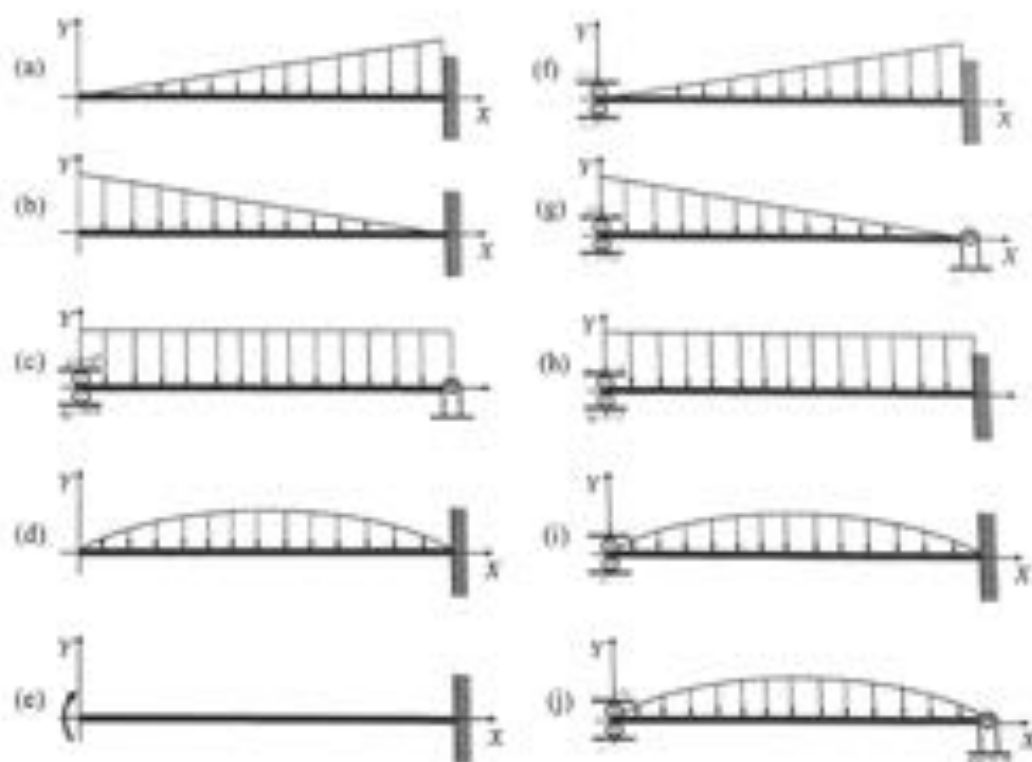
Which schematic corresponds to a beam whose bending moment is equal to

$$M(x) = \frac{w_0 x^4}{12} - \frac{w_0 L x^2}{6}$$

Circle the correct answer (a) (b) (c) (d) (e) (f) (g) (h) (i) (j)



HINT: Identify which supports and loads are compatible with the given bending moments.



P4B:

$$V(x) = \frac{dM}{dx} = \frac{w_0 L}{10} - \frac{w_0 x^2}{2L}$$
$$p(x) = \frac{dV}{dx} = \frac{-w_0 x}{L} \Rightarrow \text{linear load}$$

@x = 0

$$M(0) = 0 \quad V(0) = \frac{w_0 L}{10} \neq 0$$

Simply supported at x = 0

@x = L

$$M(L) = \frac{w_0 L^2}{10} - \frac{w_0 L^2}{6} \neq 0$$

$$V(L) = \frac{w_0 L}{10} - \frac{w_0 L}{2} \neq 0$$

Fixed end at x = L

P4C:

$$V(x) = \frac{dM}{dx} = \frac{w_0 x^3}{3} - \frac{w_0 L x^2}{2}$$

$$p(x) = \frac{dV}{dx} = w_0 x^2 - w_0 L x \Rightarrow \text{quadratic load with } p = 0 \text{ at } x = 0 \text{ and at } x = L$$

@x = 0

$$M(0) = 0 \quad V(0) = 0$$

Free end at x = 0

@x = L

$$M(L) = \frac{w_0 L^4}{12} - \frac{w_0 L^4}{6} \neq 0$$

$$V(L) = \frac{w_0 L^3}{3} - \frac{w_0 L^3}{2} \neq 0$$

Fixed end at x = L

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PROBLEM NO. 4D - 3 points max. (no partial credit)

Indicate which of the schematics presented below depicts the deflection curve of the following beam:



Circle the correct answer

(a)



(c)

(d)



(d) None of the above

PROBLEM NO. 4E - 3 points max. (no partial credit)

Indicate which of the schematics presented below depicts the deflection curve of the following beam:



Circle the correct answer

(a)

(b)



(d)



(d) None of the above