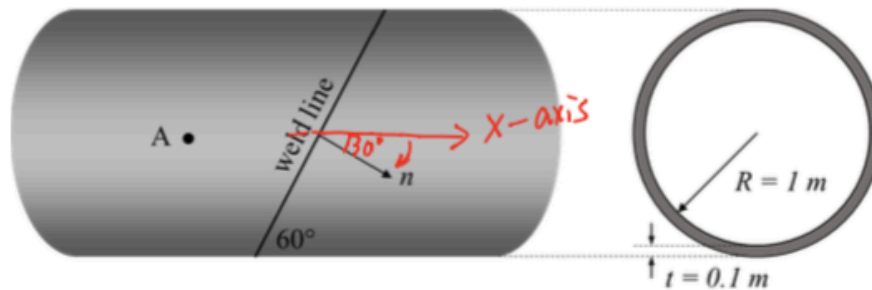


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**PROBLEM NO. 1 – 25 points max.**

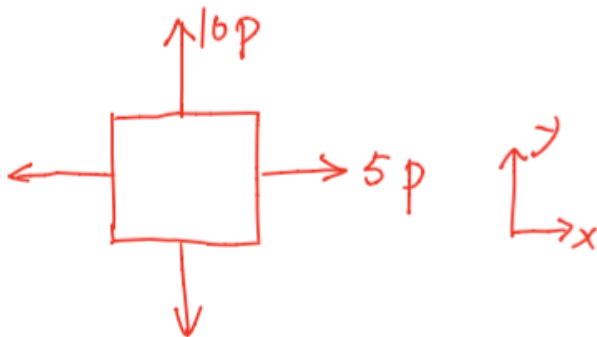
A thin-walled pressure vessel is fabricated by welding together two, open-ended stainless-steel vessels along a 60° weld line. The welded vessel has an internal radius of  $R = 1\text{ m}$  and a thickness  $t = 0.1\text{ m}$ . The gas pressure inside the vessel is  $p$ .

- Determine the axial and hoop stresses,  $\sigma_a$  and  $\sigma_h$ , in terms of the internal pressure  $p$ . Draw the state of stress at point A and its Mohr's circle.
- Use the Mohr's circle to determine the components of stress,  $\sigma_n$ ,  $\sigma_t$  and  $\tau_{nt}$ , in terms of the internal pressure  $p$ , along the weld line.
- Use the Mohr's circle to determine the maximum gas pressure  $p$ , such that BOTH of the following conditions are met:
  - The maximum shear stress of the vessel does not exceed the yield strength of stainless-steel,  $\sigma_Y = 1000\text{ MPa}$ , AND:
  - The maximum normal stress along the weld line does not exceed the welding strength in tension,  $\sigma_{wt} = 3000\text{ MPa}$ .

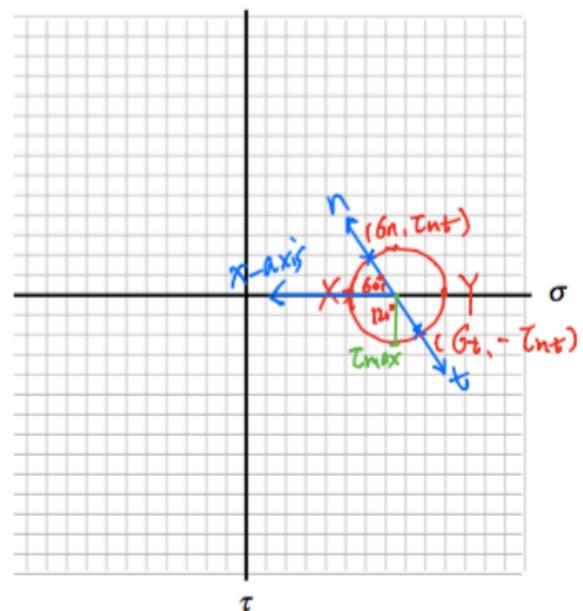


$$\sigma_a = \frac{pR}{2t} = \frac{p \cdot 1\text{ m}}{2 \times 0.1\text{ m}} = 5p$$

$$\sigma_h = \frac{pR}{t} = \frac{p \cdot 1\text{ m}}{0.1\text{ m}} = 10p$$



$$X(5p, 0) \quad Y(10p, 0)$$



$$\sigma_{ave} = \frac{\sigma_x + \sigma_y}{2} = \frac{5P + 10P}{2} = 7.5P$$

$$R = \sqrt{\left(\frac{5P - 10P}{2}\right)^2 + 0} = 2.5P$$

(2) After CW rotate  $30^\circ$

$$\sigma_n = \sigma_{ave} - R \cdot \cos 60^\circ = 7.5P - 2.5P \cdot \frac{1}{2} = 6.25P$$

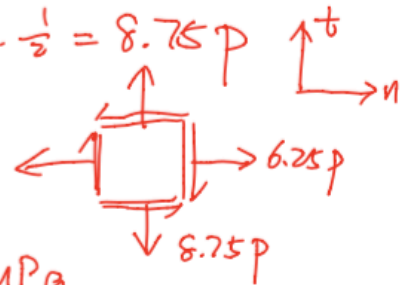
$$\tau_{nt} = -R \sin 60^\circ = -2.165P$$

$$\sigma_t = \sigma_{ave} + R \cdot \cos 60^\circ = 7.5P + 2.5P \cdot \frac{1}{2} = 8.75P$$

(3).  
I. Max. in-plan shear stress

$$\tau_{max} = R = 2.5P < 1000 \text{ MPa}$$

$$P < 400 \text{ MPa}$$



II. Max. normal stress along the weld line

$$\sigma_t = 8.75P < 3000 \text{ MPa}$$

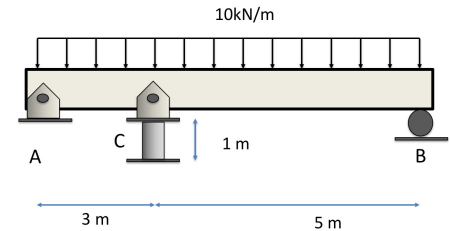
$$P < 343 \text{ MPa}$$

Max P should be 343 MPa

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**PROBLEM NO. 2 – 25 points max.**

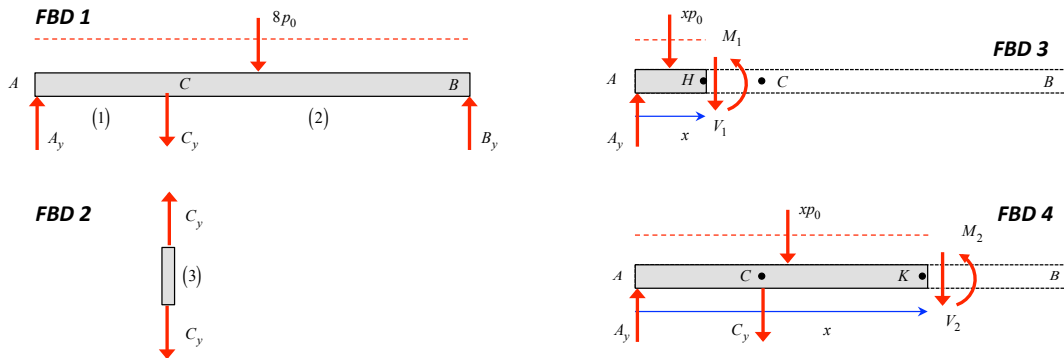
The beam AB is supported by a pin at A, a roller at B and a post having a 20 cm diameter at C. The post and the beam have a Young's modulus of  $E = 200 \text{ GPa}$ , and the second area moment of the beam is  $I = 200 \cdot 10^6 \text{ mm}^4$ . In your analysis, neglect the contribution of shear strain to the total strain energy.



- a) Using Castigliano's second theorem:
  - i. Determine the reactions on the beam at A, B and C.
  - ii. Determine the slope of the beam deflection at B.
- b) Under what assumptions can you neglect the contribution of shear strain to the total strain energy?

*If the dimensions of the beam cross section are much smaller than the length of the beam.*

SOLUTION



From FBD 1

$$\sum M_A = -C_y(3) + B_y(8) - 8p_0(4) = 0 \Rightarrow B_y = \frac{3}{8}C_y + 4p_0$$

$$\sum F_y = A_y + B_y - C_y - 8p_0 = 0 \Rightarrow A_y = \frac{5}{8}C_y + 4p_0$$

The problem is *INDETERMINATE*: will choose  $C_y$  as the redundant load.

From FBDs 3 and 4:

$$\sum M_h = -A_y x + xp_0 \left( \frac{x}{2} \right) + M_1 = 0 \Rightarrow M_1(x) = \left( 4p_0 + \frac{5}{8}C_y \right) x - \frac{1}{2}p_0 x^2$$

$$\sum M_K = -A_y x + xp_0 \left( \frac{x}{2} \right) + C_y(x-3) + M_2 = 0 \Rightarrow M_2(x) = \left( -\frac{3}{8}C_y + 4p_0 \right) x + 3C_y - \frac{1}{2}p_0 x^2$$

**METHOD #1** – consider the strain energy from only the beam ( $C_y$  is an external “applied” force)

$$U = \frac{1}{2EI} \int_0^3 M_1^2(x) dx + \frac{1}{2EI} \int_3^8 M_2^2(x) dx$$

Using Castigliano's theorem, along with rod elongation equation:  $v_C = C_y / EA$ :

$$\begin{aligned}
 -v_C &= \frac{\partial U}{\partial C_y} \Rightarrow -\frac{C_y}{EA} = \frac{1}{EI} \int_0^3 M_1 \frac{\partial M_1}{\partial C_y} dx + \frac{1}{EI} \int_3^8 M_2 \frac{\partial M_2}{\partial C_y} dx \Rightarrow \\
 0 &= \frac{1}{EI} \int_0^3 \left(\frac{5}{8}x\right) \left[ \left(4p_0 + \frac{5}{8}C_y\right)x - \frac{1}{2}p_0x^2 \right] dx + \frac{1}{EI} \int_3^8 \left(-\frac{3}{8}x + 3\right) \left[ \left(-\frac{3}{8}C_y + 4p_0\right)x + 3C_y - \frac{1}{2}p_0x^2 \right] dx + \frac{C_y}{EA} \\
 &= \frac{1}{EI} \int_0^3 \left[ \left(\frac{5}{2}p_0 + \frac{25}{64}C_y\right)x^2 - \frac{5}{16}p_0x^3 \right] dx \\
 &\quad + \frac{1}{EI} \int_3^8 \left[ 9C_y + \left(-\frac{9}{4}C_y + 12p_0\right)x + \left(\frac{9}{64}C_y - 3p_0\right)x^2 + \frac{3}{16}p_0x^3 \right] dx + \frac{C_y}{EA} \\
 &= \frac{1}{EI} \left[ \frac{1}{3} \left(\frac{5}{2}p_0 + \frac{25}{64}C_y\right)x^3 - \frac{5}{64}p_0x^4 \right]_{x=0}^{x=3} \\
 &\quad + \frac{1}{EI} \left[ 9C_yx + \frac{1}{2} \left(-\frac{9}{4}C_y + 12p_0\right)x^2 + \frac{1}{3} \left(\frac{9}{64}C_y - 3p_0\right)x^3 + \frac{3}{64}p_0x^4 \right]_{x=3}^{x=8} + \frac{C_y}{EA} \\
 &= \frac{1}{EI} \left[ \frac{1}{3} \left(\frac{5}{2}p_0 + \frac{25}{64}C_y\right)(27) - \frac{5}{64}p_0(81) \right] \\
 &\quad + \frac{1}{EI} \left[ 9C_y(8-3) + \frac{1}{2} \left(-\frac{9}{4}C_y + 12p_0\right)(64-9) + \frac{1}{3} \left(\frac{9}{64}C_y - 3p_0\right)(512-27) + \frac{3}{64}p_0(4096-81) + \frac{I}{A}C_y \right]
 \end{aligned}$$

where:  $\frac{I}{A} = \frac{(200 \times 10^6 \text{ mm}^4)(m / 1000 \text{ mm})^4}{\pi(0.2/2)^2 \text{ m}^2} = \frac{0.02}{\pi} \text{ m}^2$

Solving gives

$$C_y = -49.3 \text{ kN}$$

$$B_y = \frac{3}{8}C_y + 4p_0 = 21.5 \text{ kN}$$

$$A_y = \frac{5}{8}C_y + 4p_0 = 9.18 \text{ kN}$$

**METHOD #2** – consider the strain energy from both the beam and rod together ( $C_y$  is a reaction force)

$$U = \frac{1}{2EI} \int_0^3 M_1^2(x) dx + \frac{1}{2EI} \int_3^8 M_2^2(x) dx + \frac{1}{2} \frac{C_y^2(1)}{EA}$$

Using Castigliano's theorem:

$$0 = \frac{\partial U}{\partial C_y} = \frac{1}{EI} \int_0^3 M_1 \frac{\partial M_1}{\partial C_y} dx + \frac{1}{EI} \int_3^8 M_2 \frac{\partial M_2}{\partial C_y} dx + \frac{C_y}{EA}$$

This gives the same equation for  $C_y$  as does METHOD #1.

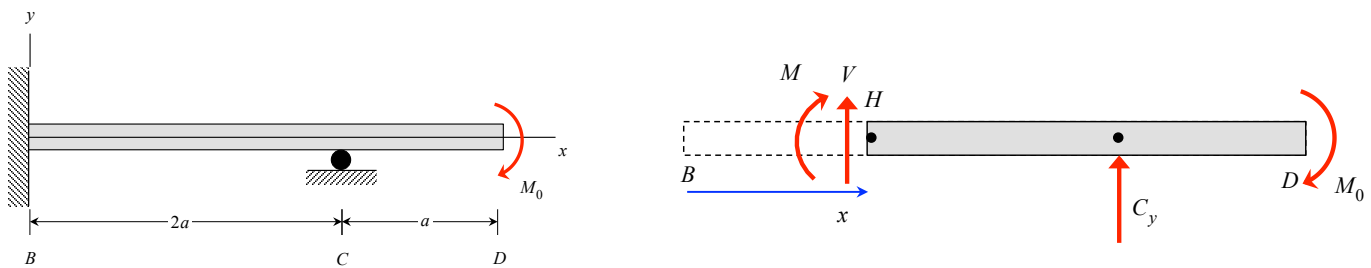
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**PROBLEM NO. 3 – 25 points max.**

Beam BD is supported by a fixed wall at end B, and by a roller support at C. A concentrated couple  $M_0$  acts at D. The beam is made up of a material with a Young's modulus  $E$  and has a constant cross section with a second area moment of  $I$  over the full length of the beam. Using either the *second-order* or *fourth-order integration* methods:

- determine the *reaction force* acting on the beam at C.
- determine the slope of the deflection  $\theta_D$  of the beam at D

Leave your answers in terms of, at most:  $M_0$ ,  $E$ ,  $I$  and  $a$ .

**Section BC**

$$\sum M_H = -M(x) + C_y(2a - x) - M_0 = 0 \Rightarrow M(x) = (2aC_y - M_0) - C_y x$$

$$\theta(x) = \theta(0) + \frac{1}{EI} \int_0^x [(2aC_y - M_0) - C_y x] dx = 0 + \frac{1}{EI} \left[ (2aC_y - M_0)x - \frac{1}{2} C_y x^2 \right]$$

$$v(x) = v(0) + \frac{1}{EI} \int_0^x \left[ (2aC_y - M_0)x - \frac{1}{2} C_y x^2 \right] dx = 0 + \frac{1}{EI} \left[ \frac{1}{2} (2aC_y - M_0)x^2 - \frac{1}{6} C_y x^3 \right]$$

Since  $v(2a) = 0$ , we have:

$$0 = \frac{1}{EI} \left[ \frac{1}{2} (2aC_y - M_0)(2a)^2 - \frac{1}{6} C_y (2a)^3 \right] = \frac{1}{EI} \left[ \frac{8}{3} C_y a^3 - 2M_0 a^2 \right] \Rightarrow C_y = \frac{3 M_0}{4 a}$$

Also,

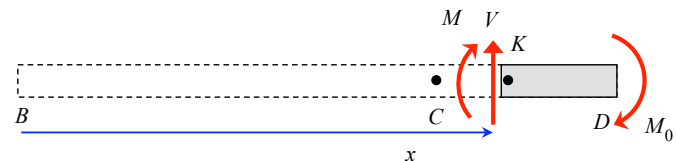
$$\theta_C = \theta(2a) = \frac{1}{EI} \left[ \left( 2a \left( \frac{3 M_0}{4 a} \right) - M_0 \right) (2a) - \frac{1}{2} \left( \frac{3 M_0}{4 a} \right) (2a)^2 \right] = \frac{1}{EI} \left[ aM_0 - \frac{3}{2} aM_0 \right] = -\frac{1}{2} \frac{aM_0}{EI}$$

**Section CD**

$$\sum M_K = -M_0 - M(x) = 0 \Rightarrow M(x) = -M_0$$

Therefore,

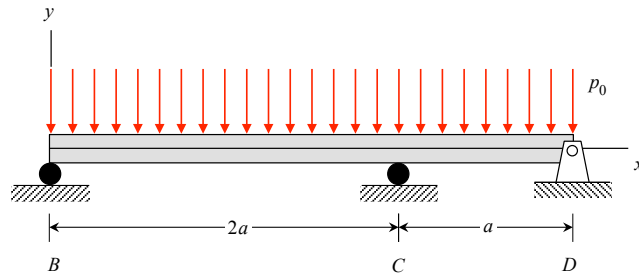
$$\theta(3a) = \theta(2a) + \frac{1}{EI} \int_{2a}^{3a} [-M_0] dx = -\frac{1}{2} \frac{M_0}{EI} a - \frac{M_0}{EI} (3a - 2a) = -\frac{3}{2} \frac{M_0}{EI} a = \theta_D$$



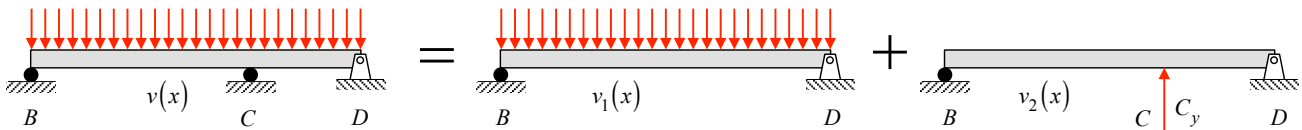
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**PROBLEM NO. 4 - PART A – 7 points max.**

A beam is made up a material with a Young's modulus of  $E$  and has a constant cross section with a second area moment of  $I$ . A downward, constant line load  $p_0$  (force/length) acts along the full length of the beam. The beam has roller supports at B and C, along with a pin joint support at end D. Using the *superposition* approach, determine the reaction force acting on the beam at the roller support C.



SOLUTION



$$v(2a) = 0 = v_1(2a) + v_2(2a)$$

$$= -\frac{1}{24}(2a) \left[ (3a)^3 - 2(3a)(2a)^2 + (2a)^3 \right] \frac{p_0}{EI} + \frac{1}{6}(a)(2a) \left[ (3a)^2 - a^2 - (2a)^2 \right] \frac{C_y}{EI(3a)}$$

$$= \frac{a^3}{EI} \left[ -\frac{11}{12} p_0 a + \frac{4}{9} C_y \right] \Rightarrow C_y = \frac{33}{16} p_0 a$$

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**PROBLEM NO. 4 - PART B – 3 points max.**

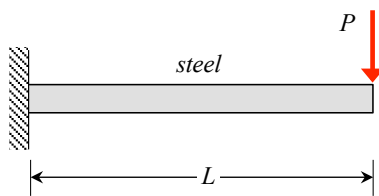
Cantilevered beams A, B and C shown below are acted upon by a point load P acting at the free end of the beam. The cross sections and lengths of each beam are the same. Beams A and B are made up of steel and aluminum, respectively, whereas Beam C is made up of both steel and aluminum components over its length. Let  $|\sigma_A|_{max}$ ,  $|\sigma_B|_{max}$  and  $|\sigma_C|_{max}$  be the maximum normal stress in Beams A, B and C, respectively. Circle the correction responses below:

TRUE or FALSE:  $|\sigma_A|_{max} = |\sigma_B|_{max}$

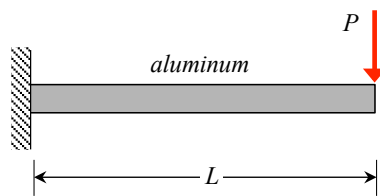
TRUE or FALSE:  $|\sigma_A|_{max} = |\sigma_C|_{max}$

TRUE or FALSE:  $|\sigma_B|_{max} = |\sigma_C|_{max}$

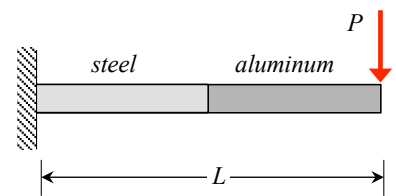
All structures are DETERMINATE.  
Stresses do not depend on materials.



Beam A



Beam B



Beam C

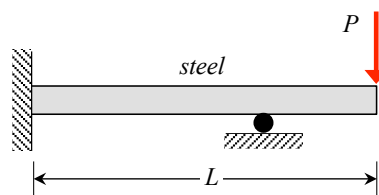
**PROBLEM NO. 4 - PART C – 2 points max.**

The propped-cantilevered beams A, B and C shown below are acted upon by a point load P acting at the free end of the beam. The cross sections and lengths of each beam are the same. Beams A and B are made up of steel and aluminum, respectively, whereas Beam C is made up of both steel and aluminum components over its length. Let  $|\sigma_A|_{max}$ ,  $|\sigma_B|_{max}$  and  $|\sigma_C|_{max}$  be the maximum normal stress in Beams A, B and C, respectively. Circle the correction responses below:

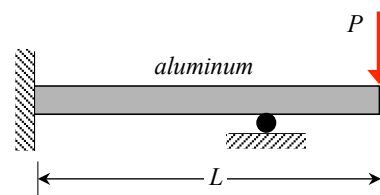
TRUE or  FALSE:  $|\sigma_A|_{max} = |\sigma_C|_{max}$

TRUE or  FALSE:  $|\sigma_B|_{max} = |\sigma_C|_{max}$

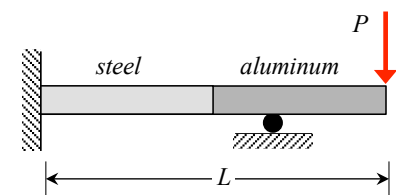
All structures are INDETERMINATE.  
Stresses depend on deflections, which are dependent on material properties.



Beam A



Beam B

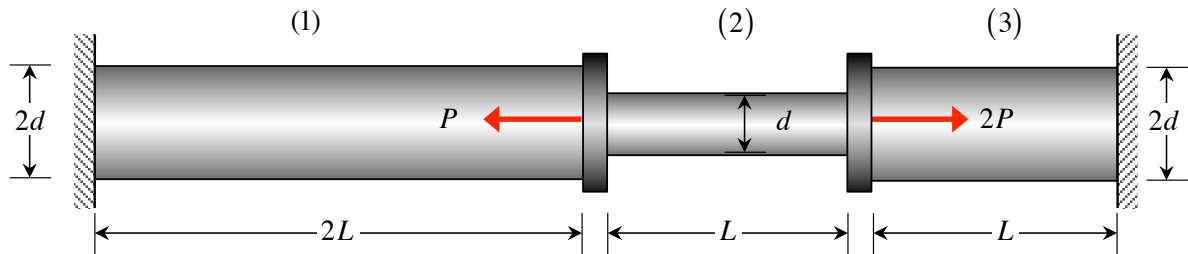


Beam C

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**PROBLEM NO. 4 - PART D – 7 points max.**

A rod is made up of solid, circular cross-section segments (1), (2) and (3), where Segment (1) has a length of  $2L$  and diameter  $2d$ , Segment (2) has a length of  $L$  and diameter  $d$ , and Segment (3) has a length of  $L$  and diameter  $2d$ . All segments are made up of a material having a Young's modulus of  $E$ . Loads of  $P$  and  $2P$  act on the rigid connectors, as shown below. You are asked to set up a three element, finite element model for displacement analysis of this rod, using one element for each of the rod segments. To this end, write down the stiffness matrix  $[K]$  and load vector  $\{F\}$  for the model *after* the boundary conditions have been enforced on the model.

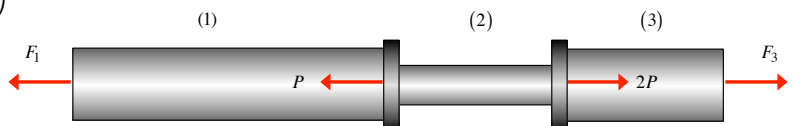


SOLUTION

$$k_1 = \frac{EA_1}{L_1} = \frac{E\pi(2d/2)^2}{2L} = \frac{\pi Ed^2}{2L} = 2\left(\pi \frac{Ed^2}{4L}\right)$$

$$k_2 = \frac{EA_2}{L_2} = \frac{E\pi(d/2)^2}{L} = \frac{\pi Ed^2}{4L} = \pi \frac{Ed^2}{4L}$$

$$k_3 = \frac{EA_3}{L_3} = \frac{E\pi(2d/2)^2}{L} = \pi \frac{Ed^2}{L} = 4\left(\pi \frac{Ed^2}{4L}\right)$$



Therefore, before enforcing BCs:

$$[K] = \pi \frac{Ed^2}{4L} \begin{bmatrix} 2 & -2 & & & \\ -2 & 3 & -1 & & \\ & -1 & 5 & -4 & \\ & & -4 & 4 & \\ & & & & \end{bmatrix}; \quad \{F\} = \begin{Bmatrix} -F_1 \\ -P \\ 2P \\ F_3 \end{Bmatrix}$$

After enforcing BCs (removing 1<sup>st</sup> and 4<sup>th</sup> rows and columns of  $[K]$  and the 1<sup>st</sup> and 4<sup>th</sup> row of  $\{F\}$ ):

$$[K] = \pi \frac{Ed^2}{4L} \begin{bmatrix} 3 & -1 \\ -1 & 5 \end{bmatrix}; \quad \{F\} = \begin{Bmatrix} -P \\ 2P \end{Bmatrix}$$



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**PROBLEM NO. 4 - PART E – 6 points max.**

A state of stress is characterized by its unknown  $x$ - $y$  components on the stress element shown below. When the stress element is rotated through an angle of  $120^\circ$ , the state of stress has the  $n$ - $t$  components shown below right.

- Draw the Mohr's circle for this state of stress on the axes provided below. Carefully label the center of the circle as well as its radius. Show the  $x$ -axis on the Mohr's circle.
- What is the maximum in-plane shear stress?
- What are the  $\sigma_x$ ,  $\sigma_y$  and  $\tau_{xy}$  components of this state of stress?

SOLUTION

$$\sigma_{ave} = \frac{\sigma_{P1} + \sigma_{P2}}{2} = \frac{22 + 12}{2} = 17 \text{ ksi}$$

$$R = \frac{\sigma_{P1} - \sigma_{P2}}{2} = \frac{22 - 12}{2} = 5 \text{ ksi}$$

$$|\tau|_{max, in-plane} = R = 5 \text{ ksi}$$

From Mohr's circle shown:

$$\sigma_x = \sigma_{ave} - R \cos 60^\circ = 17 - (5) \cos 60^\circ = 14.5 \text{ ksi}$$

$$\sigma_y = \sigma_{ave} + R \cos 60^\circ = 17 + (5) \cos 60^\circ = 19.5 \text{ ksi}$$

$$\tau_{xy} = -R \sin 60^\circ = -(5) \sin 60^\circ = -4.33 \text{ ksi}$$

