

**PROBLEM #1 (22 points)**

A solid brass core is connected to a hollow rod made of aluminum. Both are attached at each end to a rigid plate as shown in Fig. 1. The moduli of aluminum and brass are  $E_A=11,000$  ksi,  $E_B=15,000$  ksi, and the coefficients of thermal expansion are  $\alpha_A=13 \times 10^{-6}/^\circ\text{F}$ ,  $\alpha_B=12 \times 10^{-6}/^\circ\text{F}$ , respectively. The cross section areas of aluminum and brass are  $A_A=2$  in<sup>2</sup> and  $A_B=1$  in<sup>2</sup>, respectively. The core and the sleeve are stress free at the reference temperature. If the core is heated by  $100^\circ\text{F}$  and the temperature of the sleeve is held constant, calculate:

- 1) The axial force and average axial stress ( $x$  direction) in each member.
- 2) The elongation of the composite bar in the  $x$  direction.
- 3) What should be the magnitude of the force  $F$  in Fig. 1 to prevent the elongation of the composite bar in the  $x$  direction?

**Hints: Ignore the radial expansion due to Poisson's ratio effect.**

**For 1) and 2) use  $F=0$**

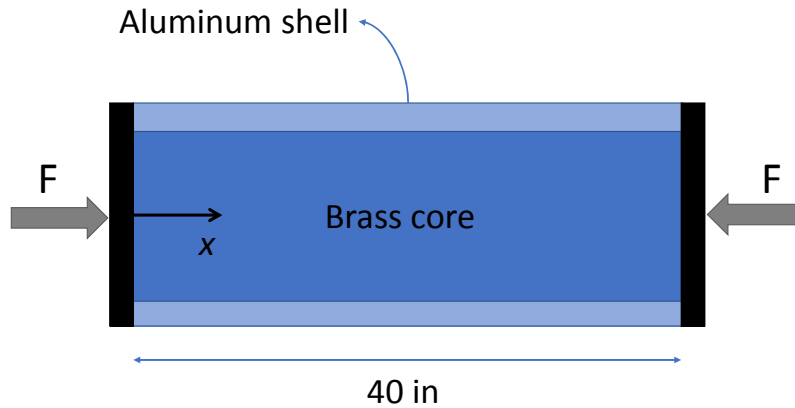
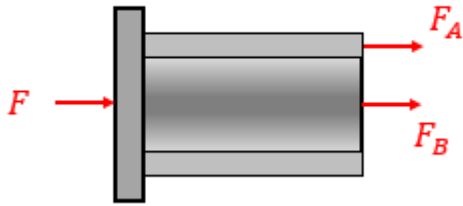


Fig. 1

**Solution:**



a)  $F = 0$

From the FBD,

$$\Sigma F_x = F_A + F_B = 0 \Rightarrow F_A = -F_B \quad (1.1)$$

From the force-elongation equation,

$$e_A = \frac{F_A L_A}{A_A E_A} \quad (1.2)$$

$$e_B = \frac{F_B L_B}{A_B E_B} + \alpha_B \Delta T L_B \quad (1.3)$$

The compatibility condition,

$$e_A = e_B \quad (1.4)$$

Combine (1.1)-(1.4) to get,

$$\begin{aligned} F_A &= 10702 \text{ lb} \\ F_B &= -10702 \text{ lb} \\ \sigma_A &= 5351 \text{ psi} \\ \sigma_B &= 10702 \text{ psi} \end{aligned}$$

b) From the compatibility equation,

$$e = e_A = e_B \quad (1.5)$$

$$e_A = \frac{F_A L_A}{A_A E_A} = 0.01946 \text{ in} \quad (1.6)$$

c) Use the same FBD,

$$\Sigma F_x = F + F_A + F_B = 0 \Rightarrow F_A + F_B = -F \quad (1.7)$$

The compatibility equation becomes,

$$e_A = e_B = 0 \quad (1.8)$$

Combine (1.2), (1.3), (1.7), (1.8) to get,

$$\begin{aligned} F_A &= \frac{e_A A_A E_A}{L_A} = 0 \text{ lb} \\ F_B &= -\alpha_B \Delta T A_B E_B = -18000 \text{ lb} \\ F &= -F_B = 18000 \text{ lb} \end{aligned}$$

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Instructor \_\_\_\_\_

**PROBLEM #2 (28 points)**

A shaft is made up of three components: solid circular shafts (1) of length  $2L$  and diameter  $d$ ; solid circular shaft (2) of length  $L$  and diameter  $2d$ ; and *tubular* shaft (3) of length  $L$ , outer diameter  $3d$  and inner diameter  $2d$ . All three components are made of the same material having a shear modulus of  $G$ . Shafts (1) and (2) are connected by a thin, rigid connector C, whereas shafts (2) and (3) are connected by a rigid connector D. Shaft (1) is rigidly attached to ground at B. Shaft (3) is rigidly attached to ground at H. Torques  $T$  and  $2T$  are applied to C and D, respectively, as shown in the Fig. 2 (a).

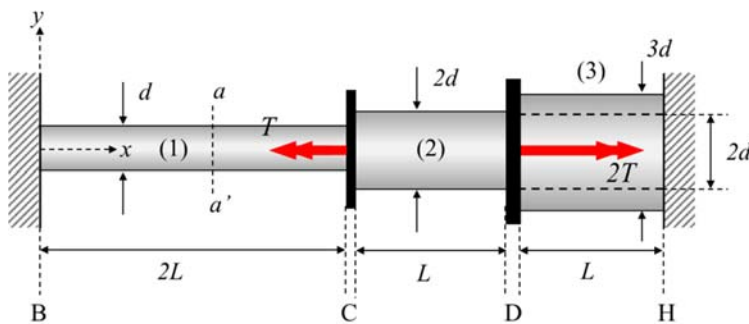
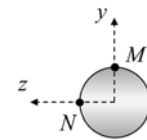
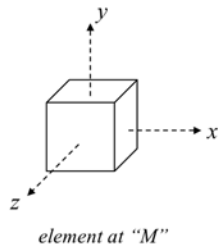


Fig. 2 (a)

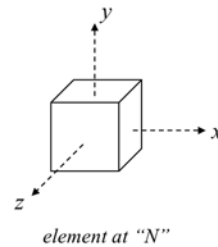


Cross-section at aa' as seen from right end

Fig. 2 (b)



element at "M"



element at "N"

Fig. 2 (c)

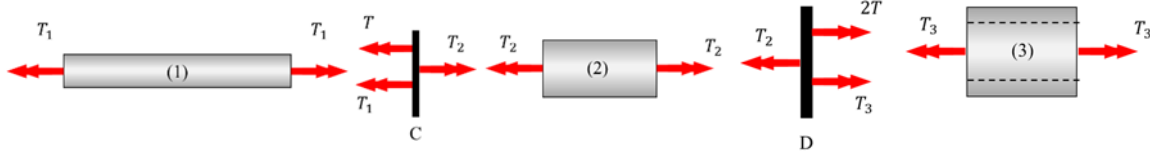
- 1) Determine the torque carried by each of the shaft components.
- 2) Determine the angle of rotation at connector C.
- 3) Show the stress element for the points M  $(x, 0.5d, 0)$  and N  $(x, 0, 0.5d)$ , whose locations on shaft (1) are also shown on Fig. 2 (b). Indicate both magnitudes and axes corresponding to the state of stress on Fig. 2 (c).

Express your final answers in terms of  $T$ ,  $L$ ,  $d$  and  $G$ .

**Solution:**

a) 1. Equilibrium.

FBD:



$$C: T_2 - T_1 - T = 0 \quad (2.1)$$

$$D: T_3 + 2T - T_2 = 0 \quad (2.2)$$

2. Torque/Rotation

$$\Delta\Phi_1 = \frac{T_1 \cdot 2L}{G \cdot Ip_1}, Ip_1 = \frac{\pi}{2} \left(\frac{d}{2}\right)^4 = \frac{\pi}{32} d^4 \Rightarrow \Delta\Phi_1 = 64 \cdot \frac{T_1 \cdot L}{G\pi d^4} \quad (2.3)$$

$$\Delta\Phi_2 = \frac{T_2 \cdot L}{G \cdot Ip_2}, Ip_2 = \frac{\pi}{2} \left(\frac{2d}{2}\right)^4 = \frac{\pi}{2} d^4 \Rightarrow \Delta\Phi_2 = 2 \cdot \frac{T_2 \cdot L}{G\pi d^4} \quad (2.4)$$

$$\Delta\Phi_3 = \frac{T_3 \cdot L}{G \cdot Ip_3}, Ip_3 = \frac{\pi}{2} \left[ \left(\frac{3d}{2}\right)^4 - \left(\frac{2d}{2}\right)^4 \right] = \frac{65}{32} \pi d^4 \Rightarrow \Delta\Phi_3 = \frac{32}{65} \cdot \frac{T_3 \cdot L}{G\pi d^4} \quad (2.5)$$

3. Compatibility

$$\Phi_C = \Phi_B + \Delta\Phi_1 = \Delta\Phi_1 \quad (\Phi_B = 0), \quad \Phi_D = \Phi_C + \Delta\Phi_2 = \Delta\Phi_1 + \Delta\Phi_2, \quad \Phi_H = \Phi_D + \Delta\Phi_3 = 0$$

$$\Rightarrow \Delta\Phi_1 + \Delta\Phi_2 + \Delta\Phi_3 = 0 \quad (2.6)$$

$$\text{From (2.3) to (2.6), we have } 64 \cdot \frac{T_1 \cdot L}{G\pi d^4} + 2 \cdot \frac{T_2 \cdot L}{G\pi d^4} + \frac{32}{65} \cdot \frac{T_3 \cdot L}{G\pi d^4} = 0$$

$$\Rightarrow 64T_1 + 2T_2 + \frac{32}{65}T_3 = 0 \quad (2.7)$$

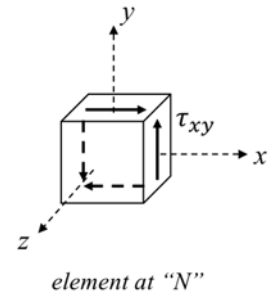
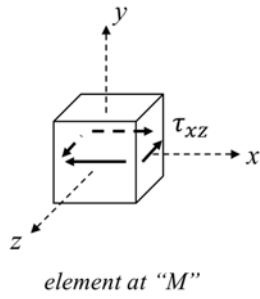
4. Solve

With (2.1), (2.2) and (2.7), we can solve

$$\begin{cases} T_1 = -0.023T \\ T_2 = 0.977T \\ T_3 = -1.023T \end{cases}$$

$$b) \Phi_C = \Delta\Phi_1 = 64 \cdot \frac{T_1 \cdot L}{G\pi d^4} = 64 \cdot \frac{-0.023T \cdot L}{G\pi d^4} = -0.468 \frac{T \cdot L}{Gd^4} \text{ (radians)}$$

c) Since the elements M and N are both at the circumference of the cross section at aa', and a negative torque ( $T_1$ ) has been applied, their states of stress are shown as below:

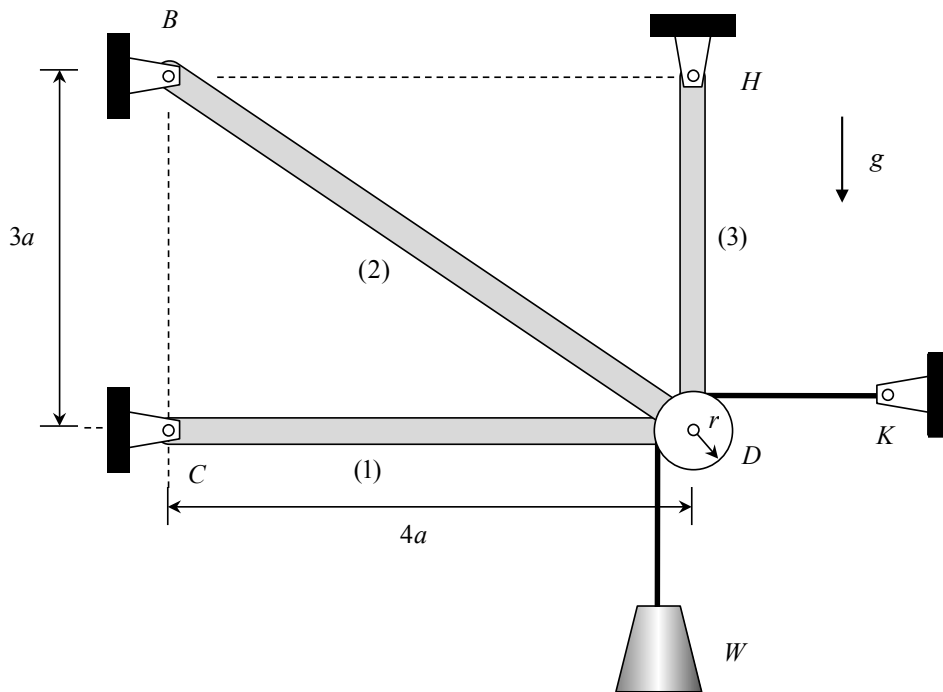


The magnitude of shear stresses  $|\tau_{xz}| = |\tau_{xy}| = \left| \frac{T_1 \cdot \frac{d}{2}}{Ip_1} \right| = \left| \frac{-0.023T \cdot \frac{d}{2}}{\frac{\pi d^4}{32}} \right| = 0.117 \frac{T}{d^3}$

**PROBLEM #3 (26 points)**

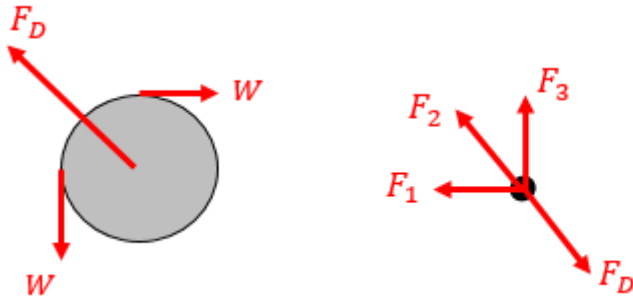
A truss is made up of members (1)-(3), with each member having a cross-sectional area of  $A$  and being made up of a material with a Young's modulus of  $E$ . Members (1)-(3) are connected to ground at pins C, B and H, respectively, and are connected together at pin D. A frictionless pulley is also attached to pin D. A block of weight  $W$  is supported by a cable that is pulled over the pulley and attached to ground at K. Pins B, C and H each have a cross-sectional diameter of  $d$ , and have a "single-sided" connection to ground. The pins are made of a material having a shear yield strength of  $\tau_Y$ .

- Determine the load carried by each member of the truss. Express your final answers in terms of, at most,  $A$ ,  $E$ ,  $W$  and  $a$ .
- Determine the largest  $W$  that can be supported by the truss without failure at pins B, C or H. Use a factor of safety of  $FS=2$ .



**Solution:**

a)



From the FBD,

$$F_D = \sqrt{2}w$$

$$\Sigma F_x = 0 \Rightarrow F_1 + 0.8F_2 = w \quad (3.1)$$

$$\Sigma F_y = 0 \Rightarrow 0.6F_2 + F_3 = w \quad (3.2)$$

For each truss,

$$\theta_1 = 0, \quad \theta_2 = 323^\circ, \quad \theta_3 = 270^\circ, \quad (3.3)$$

From the force-elongation equation,

$$e_1 = \frac{F_1 L_1}{E_1 A_1} = \frac{F_1 4a}{EA} \quad (3.4)$$

$$e_2 = \frac{F_2 L_2}{E_2 A_2} = \frac{F_2 5a}{EA} \quad (3.5)$$

$$e_3 = \frac{F_3 L_3}{E_3 A_3} = \frac{F_3 3a}{EA} \quad (3.6)$$

Compatibility conditions,

$$e_1 = u_D \cos \theta_1 + v_D \sin \theta_1 = u_D \quad (3.7a)$$

$$e_2 = u_D \cos \theta_2 + v_D \sin \theta_2 = 0.8u_D - 0.6v_D \quad (3.7b)$$

$$e_3 = u_D \cos \theta_3 + v_D \sin \theta_3 = -v_D \quad (3.7c)$$

From (3.7a) to (3.7c) to get,

$$e_2 = 0.8e_1 + 0.6e_3 \quad (3.8)$$

Combine (3.4) to (3.6) with (3.8) to get,

$$5F_2 = \frac{16}{5}F_1 + \frac{9}{5}F_3 \quad (3.9)$$

Solve (3.1), (3.2) and (3.9) to get,

$$F_1 = 0.537w$$

$$F_2 = 0.578w$$

$$F_3 = 0.653w$$

b) From the failure theory,

$$FS = \frac{\tau_y}{\tau_{\text{allow}}}$$

The problem is for the single pin situation,

$$V = F$$

For point B,

$$\tau_B = \frac{F_2}{A} = \frac{0.578w}{\frac{\pi d^2}{4}} = \frac{\tau_y}{2} \Rightarrow (w_B)_{\text{allow}} = 0.678\tau_y d^2$$

For point C,

$$\tau_c = \frac{F_1}{A} = \frac{0.537w}{\frac{\pi d^2}{4}} = \frac{\tau_y}{2} \Rightarrow (w_c)_{\text{allow}} = 0.731\tau_y d^2$$

For point H,

$$\tau_H = \frac{F_3}{A} = \frac{0.653w}{\frac{\pi d^2}{4}} = \frac{\tau_y}{2} \Rightarrow (w_H)_{\text{allow}} = 0.601\tau_y d^2$$

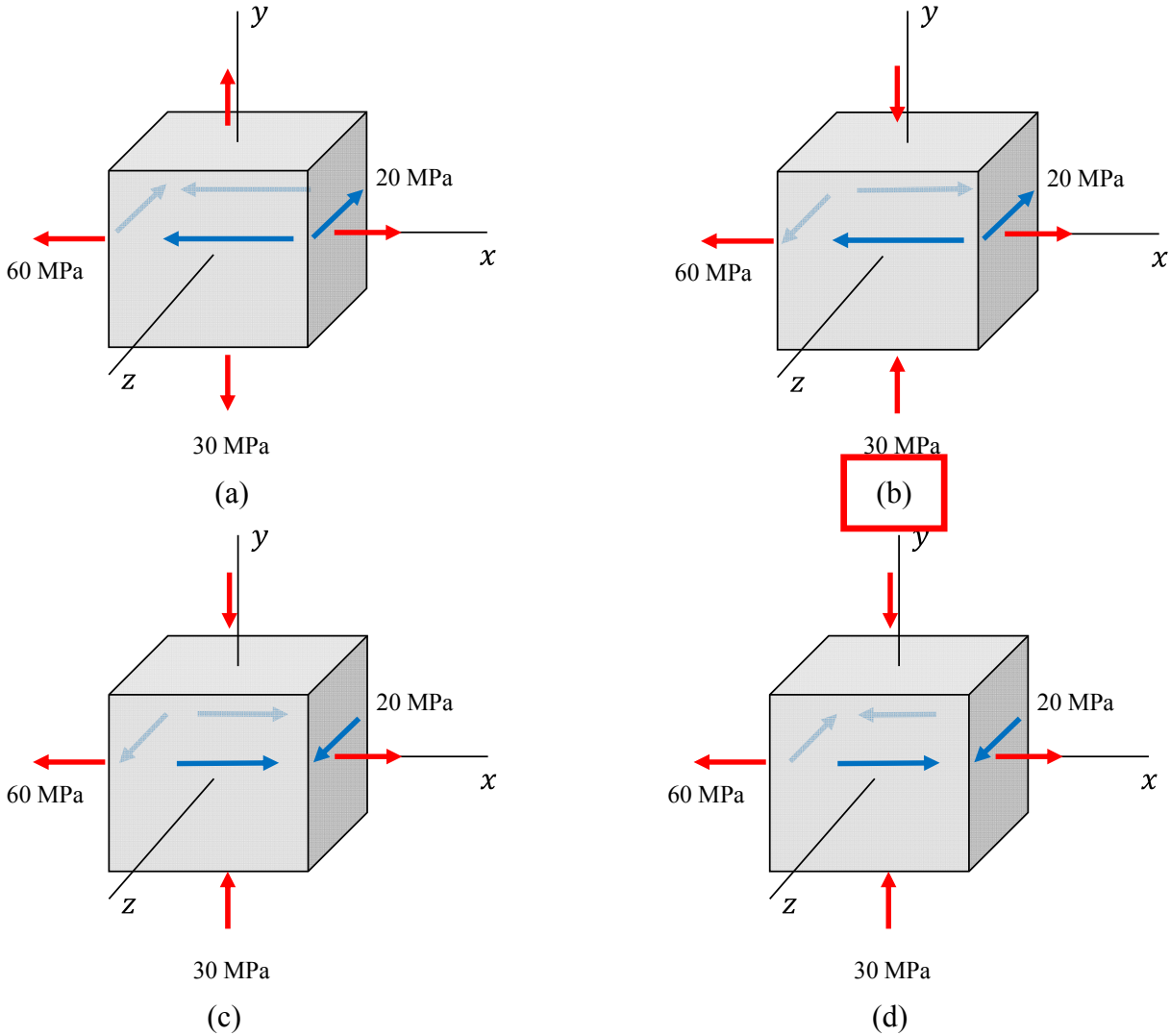
Thus,

$$w_{\text{allow}} = \text{smallest } w = (w_H)_{\text{allow}} = 0.601\tau_y d^2$$



**PROBLEM #4 (24 Points):**

4.1. A material point in a steel machine is subjected to the following stress state:  $\sigma_x = 60 \text{ MPa}$ ,  $\sigma_y = -30 \text{ MPa}$ , and  $\tau_{xz} = -20 \text{ MPa}$ . Which of the following stress elements represents the correct stress state of the material point?



Using  $E=210 \text{ GPa}$  and  $\nu=0.3$  for steel,  $G = \frac{E}{2(1+\nu)}$ , determine the strain components:

$$\epsilon_x = \frac{\sigma_x - \nu\sigma_y}{E} = 0.00033$$

$$\epsilon_y = \frac{\sigma_y - \nu\sigma_x}{E} = -0.00023$$

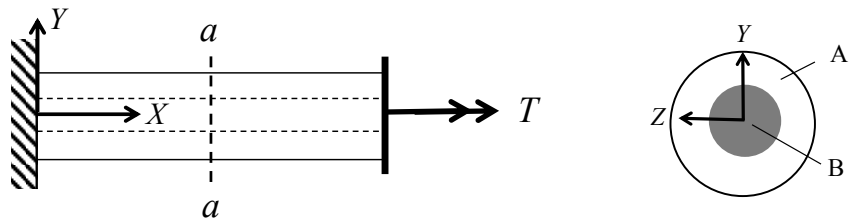
$$\epsilon_z = \frac{-\nu(\sigma_x + \sigma_y)}{E} = -0.00043$$

$$\gamma_{xy} = \frac{\tau_{xy}}{G} = 0$$

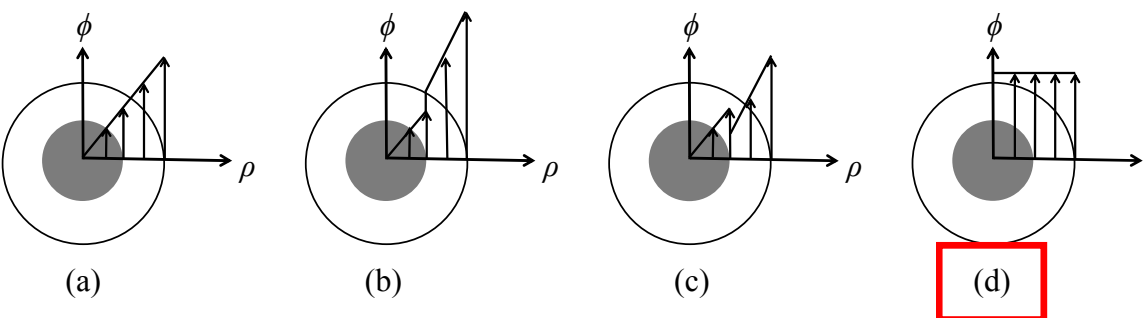
$$\gamma_{yz} = \frac{\tau_{yz}}{G} = 0$$

$$\gamma_{xz} = \frac{\tau_{xz}}{G} = -0.00025$$

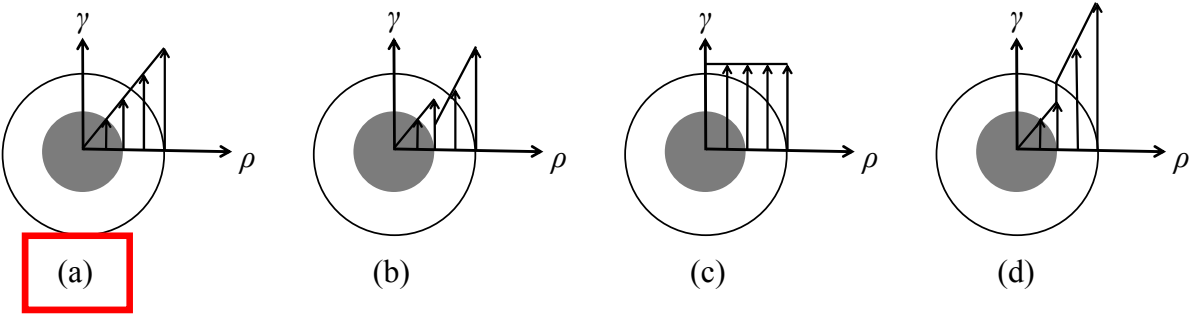
**4.2.** A bimetallic bar with circular cross section consists of a shell A and a core B. The bimetallic bar is subjected to a torque  $T$ . The shear moduli of the core and shell are known to be  $G_A = 2G_B$ , and polar moment of inertia  $I_{PA} = 0.5 I_{PB}$ .



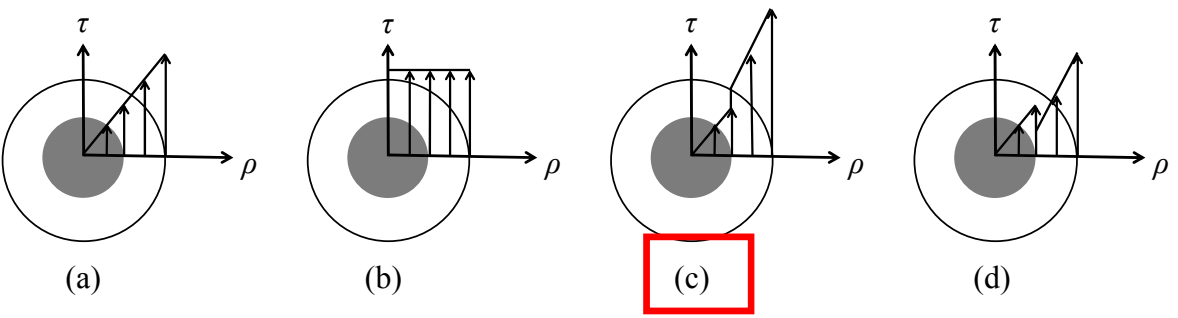
Which figure shows the correct distribution of the twist angle in the cross section  $aa$ ?



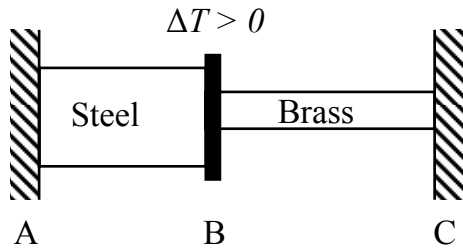
Which figure shows the correct distribution of the shear strain in the cross section  $aa$ ?



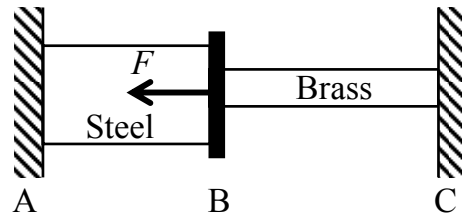
Which figure shows the correct distribution of the shear stress in the cross section  $aa$ ?



4.3. A stepped rod is composed of steel and brass which are joined by a rigid connector at B and are fixed at the two ends. The coefficients of thermal expansion for steel and brass are  $\alpha_S$  and  $\alpha_B$  ( $\alpha_S > \alpha_B$ ), respectively. Consider two load conditions shown below, the rod is homogeneously heated with a temperature increase of  $\Delta T$  in (1), and a force  $F$  is applied at the rigid connector while temperature is held constant in (2). Which of the following statements are correct? (More than one item can be selected)



(1)



(2)

- (a) Both configurations (1) and (2) are statically indeterminate structures.
- (b) (1) is a statically determinate structure and (2) is a statically indeterminate structure.
- (c) The forces  $F_S$  (steel) and  $F_B$  (brass) in (1) are both zero
- (d) The forces  $F_S$  (steel) and  $F_B$  (brass) in (1) are non-zero and are equal.
- (e) The forces  $F_S$  (steel) and  $F_B$  (brass) in (1) are equal in magnitude and of opposite signs.
- (f) The strains  $\epsilon_S$  (steel) and  $\epsilon_B$  (brass) in (1) are both zero.
- (g) The strains  $\epsilon_S$  (steel) and  $\epsilon_B$  (brass) in (1) are non-zero and are equal.
- (h) The strains  $\epsilon_S$  (steel) and  $\epsilon_B$  (brass) in (1) are equal in magnitude and of opposite signs.
- (i) The stresses  $\sigma_S$  (steel) and  $\sigma_B$  (brass) in (1) are both compressive.
- (j) The stresses  $\sigma_S$  (steel) and  $\sigma_B$  (brass) in (1) are equal in magnitude and of opposite signs.
- (k) The forces  $F_S$  (steel) and  $F_B$  (brass) in (2) are non-zero and are equal.
- (l) The forces  $F_S$  (steel) and  $F_B$  (brass) in (2) are both compressive.
- (m) The elongations  $e_S$  (steel) and  $e_B$  (brass) in (2) are equal.
- (n) The elongations  $e_S$  (steel) and  $e_B$  (brass) in (2) are equal in magnitude and of opposite signs.
- (o) The stresses  $\sigma_S$  (steel) and  $\sigma_B$  (brass) in (2) are equal in magnitude and of opposite signs.
- (p) The solution of the stresses  $\sigma_S$  (steel) and  $\sigma_B$  (brass) in (2) depends on the cross sections, lengths, and elastic moduli of steel and brass.