<Solution for Problem 1>

(a)

+2 Air at 350K 
$$\Rightarrow$$
  $v_{air} = 20.92 \times 10^{-6} m^2/s$ ,  $Pr_{air} = 0.7$ ,  $k_{air} = 30 \times 10^{-3} W/m - K$ 

+3 
$$\operatorname{Re}_{D} = \frac{V \cdot D}{V} = \frac{4 \ m/s \times 0.04 \ m}{20.92 \times 10^{-6} \ m^{2}/s} = 7646.49$$

$$\underline{+3} \qquad \overline{Nu_D} = 0.3 + \frac{0.62 \operatorname{Re}_D^{\frac{1}{2}} \operatorname{Pr}^{\frac{1}{3}}}{\left[1 + (0.4 / \operatorname{Pr})^{\frac{2}{3}}\right]^{\frac{1}{4}}} \cdot \left[1 + \left(\frac{\operatorname{Re}_D}{28,2000}\right)^{\frac{5}{8}}\right]^{\frac{4}{5}} = 46.04$$

$$\frac{+2}{h} = \frac{k}{D} \cdot \overline{Nu_D} = \frac{30 \times 10^{-3} \ W/m - K}{0.04 \ m} \cdot (46.04) = 34.53 \ W/m^2 - K$$

$$\therefore \ \overline{h} = 34.53 \ W / m^2 - K$$

$$\underline{+4} \quad Bi = \frac{h \cdot L_c}{k_g} = \frac{h \cdot R}{k_g} = \frac{(34.53 \ W/m^2 - K) \cdot (0.02 \ m)}{1.38 \ W/m - K} = 0.5$$

For Bi = 0.5,  $\zeta_1 = 0.9408$  and  $C_1 = 1.1143$  (cylinder)

$$\frac{+3}{R^2} \quad Fo = \frac{\alpha \cdot t}{R^2} = \frac{\left(0.8 \times 10^{-6} \ m^2/s\right) \cdot (120 \ s)}{\left(0.02 \ m\right)^2} = 0.24 > 0.2$$

$$\frac{+3}{T_i - T_{\infty}} = \frac{T_{r=0,t=120s} - T_{\infty}}{T_i - T_{\infty}} = C_1 \cdot \exp\left(-\zeta_1^2 \cdot Fo\right) = 1.1143 \cdot \exp\left(-0.9408^2 \cdot 0.24\right) = 0.901046$$

$$\underline{+2} \quad T_{r=0,t=120s} = T_{\infty} + \theta_0^* \cdot (T_i - T_{\infty}) = 300 \ K + 90.1 \ K = 390.1 \ K$$

## $\therefore T_{r=0,t=120s} = 390.1 K$

(c) <u>+8</u>



No Assumptions -1

<Solution for Problem 2>

(a)

+3 
$$q_{electric} = q_{conv} + q_{evap}$$

$$\underline{+4} \quad q_{conv} = h \cdot A \cdot (T_s - T_{\infty})$$

$$\underline{+4} \quad q_{evap} = \dot{m} \cdot h_{fg} = n_A^{"} [\rho_{A,sat}(T_s) - \rho_{A,\infty}(T_{\infty})] \cdot A \cdot h_{fg}$$

where 
$$\dot{m} = n_A^{"} \cdot A = h_m \cdot A \cdot \left[ \rho_{A,sat}(T_s) - \rho_{A,\infty}(T_{\infty}) \right]$$

Overall,

$$q_{electric} = \frac{\dot{m} \cdot (T_s - T_{\infty})}{\rho_{A,sat}(T_s) - \rho_{A,\infty}(T_{\infty})} \cdot \rho c_p L e^{2/3} + \dot{m} \cdot h_{fg}$$

$$\pm 1 \qquad \therefore q_{electric} = q_{conv} + q_{evap} \text{ or } q_{electric} = \frac{\dot{m} \cdot (T_s - T_{\infty})}{\rho_{A,sat}(T_s) - \rho_{A,\infty}(T_{\infty})} \cdot \rho c_p L e^{2/3} + \dot{m} \cdot h_{fg}$$

$$\underline{+4} \quad \frac{h}{h_m} = \frac{h \cdot A}{h_m \cdot A} = \rho c_p L e^{2/3}$$

$$h \cdot A = h_m \cdot A \cdot \rho c_p L e^{2/3} = \frac{\dot{m}}{\rho_{A,sat}(T_s) - \rho_{A,\infty}(T_{\infty})} \cdot \rho c_p L e^{2/3}$$

<u>+4</u> where  $Le = \frac{\alpha}{D_{AB}}$ 

$$\underline{+4} : h \cdot A = \frac{\dot{m}}{\rho_{A,sat}(T_s) - \rho_{A,\infty}(T_{\infty})} \cdot \rho c_p L e^{2/3}$$

(c)

$$\begin{array}{rcl} \pm 1 & T_{film} &= \frac{T_s + T_{\infty}}{2} &= \frac{(310 + 290) \ K}{2} &= 300 \ K \\ & At \ 300 \ K, \\ \pm 1 & \rho_{air} &= 1.1614 \ kg/m^3, \ c_{p,air} &= 1,007 \ J/kg - K, \ \alpha_{air} &= 22.5 \times 10^{-6} \ m^2/s, \ D_{AB} &= 26 \times 10^{-6} \ m^2/s \\ & At \ T_s &= 310 \ K, \\ \pm 1 & \rho_{A,s} &= \frac{1}{v_{g,air}} &= \frac{1}{22.93} \ m^3/kg \ = \ 0.043611 \ kg/m^3, \ h_{fg} \ = \ 2,414 \times 10^3 \ J/kg \end{array}$$

+1 Also, 
$$\rho_{A,\infty} = 0$$
 (dry air)

$$\underbrace{\mathbf{+1}}_{\mathbf{+1}} \quad q_{electric} = \frac{\dot{m} \cdot (T_s - T_{\infty})}{\rho_{A,sat}(T_s) - \rho_{A,\infty}(T_{\infty})} \cdot \rho_{air} c_{p,air} Le^{2/3} + \dot{m} \cdot h_{fg} = \frac{\dot{m} \cdot (T_s - T_{\infty})}{\rho_{A,sat}(T_s)} \cdot \rho_{air} c_{p,air} \left(\frac{\alpha}{D_{AB}}\right)^{2/3} + \dot{m} \cdot h_{fg}$$

$$\therefore \quad q_{electric} = \frac{\dot{m} \cdot (310 - 290)K}{0.043611 \ kg/m^3} \cdot (1.1614 \ kg/m^3) \cdot (1,007 \ J/kg - K) \left(\frac{22.5 \times 10^{-6} \ m^2/s}{26 \times 10^{-6} \ m^2/s}\right)^{2/3} + \dot{m} \cdot (2,414 \times 10^3 \ J/kg) = 3,000 \ W$$

$$\Rightarrow \quad \dot{m} = 0.001034 \ kg/s$$

$$\pm 1$$
  $\therefore$   $\dot{m} = 0.001034 \ kg/s$ 

No Assumptions -1

<Solution for Problem 3>

(a)

At 
$$T_{film,i} = 162.5^{\circ}C$$
,  $v_{air} = 30.67 \times 10^{-6} m^2/s$ ,  $Pr_{air} = 0.687$ ,  $k_{air} = 0.0363 W/m-K$ 

$$\frac{\pm 3}{2} \operatorname{Re}_{L} = \frac{U_{\infty}L}{v} = \frac{(30 \text{ m/s}) \cdot (1 \text{ m})}{30.67 \times 10^{-6} \text{ m}^{2}/\text{s}} = 9.781155 \times 10^{5} > \operatorname{Re}_{critical} = 5 \times 10^{5} \Rightarrow \text{mixed boundary layer}$$

$$\frac{\pm 3}{10^{5}} = 9.781155 \times 10^{5} \Rightarrow \text{mixed boundary layer}$$

(b)

$$\underline{+4} \quad \overline{Nu_L} = (0.037 \cdot \text{Re}_L^{4/5} - 871) \cdot \text{Pr}^{1/3} = [0.037 \cdot (9.78155 \times 10^5)^{4/5} - 871] \cdot (0.687)^{1/3} = 1,255.31$$

$$\underline{+4} \quad \overline{Nu_L} = \frac{\overline{h_L}L_c}{k_{air}} = 1,255.31 \implies \overline{h_L} = 45.5678 \quad W/m^2 - K$$

$$\boxed{\therefore \overline{h_L}} = 45.5678 \quad W/m^2 - K$$

$$\underline{+3} \quad Bi = \frac{\overline{h_L} \cdot L_c}{k_s} \quad where \quad L_c = \frac{V}{A_s} = \frac{D}{4}$$

$$\underline{+2} \quad Bi = \frac{\left(45.5678 \ W/m^2 - K\right) \cdot \left(\frac{0.1 \ m}{4}\right)}{100 \ W/m - K} = 0.011392 < 0.1 \Rightarrow Lump \ capacitance \ analysis \ is \ valid.$$

 $\frac{\pm 3}{\therefore Bi = 0.011392 \Rightarrow Lump \ capacitance \ analysis \ is \ valid.}$ 

(d)

$$\underline{+4} \quad \frac{\theta}{\theta_i} = \frac{T - T_i}{T_{\infty} - T_i} = \frac{(30 - 25)^{\circ}C}{(300 - 25)^{\circ}C} = 0.018182 = \exp(-Bi \cdot Fo)$$

where 
$$Fo = \frac{\alpha \cdot t}{L_c^2}$$
 and  $\alpha = \frac{k_s}{\rho_s \cdot c_{p,s}} = \frac{100 \ W/m - k}{(2,702 \ kg/m^3) \cdot (903 \ J/kg - K)} = 4.0985 \times 10^{-5} \ m^2/s$ 

$$\underline{+2} \quad \therefore \frac{\theta}{\theta_i} = 0.018182 = \exp\left[-(0.011392) \cdot \frac{\left(4.0985 \times 10^{-5} \ m^2/s\right) \cdot t_{30^\circ C}}{\left(\frac{0.1 \ m}{4}\right)^2}\right]$$

<u>+4</u> Solve for  $t_{30^\circ C} \Rightarrow t_{30^\circ C} = 5,364.28$  seconds

$$\therefore t_{30^{\circ}C} = 5,364.28 \ seconds = 1.49 \ hrs.$$

(e)

 $Re_{x=10cm} < Re_{x,critical} \Rightarrow$  The turbulent transition is tripped earlier than without the ring case



For laminar flow,  $h \sim x^{-1/2}$ , while for turbulent flow  $h \sim x^{-1/5}$ .

- Hence h varies more slowly for turbulent flow, and also has larger magnitude at given location, x.
- Thus, by having a larger portion of the surface as turbulent (*e.g.* by tripping the boundary layer at Re<sub>x</sub> < Re<sub>x,c</sub>), *h<sub>L</sub>* will be larger.

Correct answer (greater heat transfer) +1